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Abstract

The effect of two thermal barrier coatings and their surface roughness on heat transfer, combustion, and emissions has been investigated in a single-cylinder light-duty diesel engine. The evaluated thermal barrier coating materials were plasma-sprayed yttria-stabilized zirconia and hard anodized aluminum, which were applied on the piston top surface. The main tool for the investigation was cylinder pressure analysis of the high-pressure cycle, from which the apparent rate of heat release, indicated efficiency, and heat losses were derived. For verification of the calculated wall heat transfer, the heat flow to the piston cooling oil was measured as well. Application of thermal barrier coatings can influence engine operating conditions like charge temperature and ignition delay. Therefore, extra attention was paid to choosing stable and repeatable engine operating points. The experimental data were modeled using multiple linear regression to isolate the effects of the coatings and of the surface roughness. The results from this study show that high surface roughness leads to increased wall heat losses and a delayed combustion. However, these effects are less pronounced at lower engine loads and in the presence of soot deposits. Both thermal barrier coatings show a reduction of cycle-averaged wall heat losses, but no improvement in indicated efficiency. The surface roughness and thermal barrier coatings had a significant impact on the hydrocarbon emissions, especially for low-load engine operation, while their effect on the other exhaust emissions was relatively small.

Keywords

Engine, efficiency, emissions, thermal barrier coating, TBC, surface roughness, MLR, regression, data model

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Introduction

The continuous need for reduction of fuel consumption and CO₂ emissions from road transportation has led to a renewed interest in thermal insulation of the combustion chamber in the internal combustion engine. The heat loss to the combustion chamber walls is one of the major energy losses in an engine. One way to reduce this heat loss is the application of a thermal barrier coating (TBC).

In theory, thermal insulation of the combustion chamber has a range of benefits: increased indicated efficiency, lower heat load on critical engine components, and a smaller and cheaper cooling system. Effective insulation would also increase the exhaust gas temperature, which is useful in two ways: (1) the increase in enthalpy can reduce pumping losses when using a modern turbocharging system or it can be used in a waste heat recovery system. (2) The higher exhaust temperature improves the conversion efficiency in a

catalytic exhaust gas aftertreatment system, especially for light engine loads (city driving) and cold ambient conditions.

Thermal insulation of the combustion chamber with TBCs has been investigated since the early 1980s by many researchers, with experiments as well as thermodynamic process simulations. Most of the experiments show an increase in exhaust temperature and a reduction in heat losses to the coolant, as expected and predicted by simulations. However, the experimental results for the engine efficiency show a large variation, and there is no general agreement on the benefit of TBCs for indicated efficiency.^{1–3} Moreover, it seems

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that, on average, the effect of insulation with TBCs on indicated efficiency is limited.

The variation in the reported benefits from TBCs is partly related to the large range of tested engine hardware, the variation in engine operating conditions, and the engine technology level used in the experiments. Another cause for the varying results from experiments is the increased wall temperature that follows with insulation. This temperature increase results in a higher charge temperature and lower charge density. For compression ignition (CI) engines, this can lead to a lower air–fuel ratio and differences in ignition delay, fuel–air mixing, emission formation, and oxidation. Especially, combustion phasing and the rate of heat release have a significant effect on the indicated efficiency. How these secondary effects change the efficiency is engine-specific and depending on engine load. Kobori et al.⁴ published a detailed overview of these secondary effects for an insulated CI engine.

The limited effect of TBCs on indicated efficiency has been addressed by several researchers. One of the proposed explanations is that high surface roughness, which is typical for plasma-sprayed and anodized TBCs, increases heat transfer and slows down combustion. These negative effects from high surface roughness have been reported in publications on spark ignition^{5,6} as well as CI engines.^{7,8} To mitigate the negative impact of surface roughness in a CI engine, Kawaguchi et al.⁹ limited the application of their new TBC to the piston top surface, excluding the bowl. The typical reported difference in engine efficiency is about 1%–3% between a smooth and rough surface finish. However, in some cases, no effect was shown, or efficiency deteriorated as much as 6%.

The published experimental results, particularly for CI engines, do not show a clear consensus with respect to the effectiveness of TBCs and the negative impact of surface roughness on indicated efficiency.

In this article, a robust and comprehensive method is presented for experimental investigation of TBCs and surface roughness in CI engines. The boundary conditions were accurately controlled, and the method is insensitive for secondary effects of the intake air heating. A complete set of result data is included, comprising energy balance, combustion analysis and emissions for low-, medium-, and high-load engine operating conditions. Variations in emissions can give insight in changes in the combustion process. The method is applied on two representative TBCs with a typically high surface roughness. The location of the coatings and the level of surface roughness are varied to isolate and quantify the effect of surface roughness and the coatings on engine performance.

This article is organized as follows: In the “Method” section, the measurement robustness improvements are presented as well as the equations for the apparent heat release rate and energy balance. A description of the used statistical method ends the “Method” section. The section about the experimental setup describes the test engine

and measurement equipment as well as the properties of the tested pistons and the definition of the test sequence. The results of the engine experiments are then presented in three main parts: first, the apparent rate of heat release (aRoHR) is compared and analyzed for the different pistons; in the second part, the energy balance and emissions for fired operation are shown, and finally, the energy balance and impact of soot deposits for motored engine operation are discussed. Before ending with the conclusion, a table is presented that summarizes the effects of the coatings and surface roughness on the indicated efficiency, heat losses, and engine emissions.

Method

Measurement robustness

A number of measures were taken to make the measurements as accurate as possible and to reduce the secondary effects of the application of the TBCs, such as reduced volumetric efficiency and changes of ignition delay. The temperature levels for coolant, engine oil, fuel supply, and intake air were controlled within ± 0.1 °C. Intake pressure and exhaust back-pressure were controlled within ± 1 kPa. The target setting for the injected fuel quantity was fixed for each engine operating point (EOP). In this way, changes in energy distribution could be investigated, without changing the conditions for combustion such as the air–fuel ratio. Furthermore, the following measures were applied:

- A single pilot injection of 2 mg/stroke was used to stabilize the ignition timing of the main combustion.
- Each test run was started with clean combustion chamber surfaces; cleaned or first-run engine parts.
- An automated test sequence was applied to target identical time and temperature histories for the different pistons.
- Motored EOPs were included for thermodynamic assessment of the actual compression ratio and to verify proper functioning of the indicating equipment through the test run.
- The intake air mass flow was kept constant by adjusting the intake air temperature.

By controlling the intake temperature to target a constant air mass flow, the in-cylinder conditions at intake valve closing should have been similar for all tested pistons. The impact of TBCs on intake charge heating could still be assessed by the volumetric efficiency, which is related to the temperature in the intake manifold.

Calculation of heat release and energy balance

The analysis of the combustion process and the energy balance in this article is largely based on data from cylinder pressure measurements. This measurement is relatively simple and gives information of time-resolved

processes and complete cycle energy flows. The measurement of heat loss to the piston cooling oil was added to complement the thermodynamic calculation of the heat losses with the purpose to validate the calculated heat loss and to see possible variations in heat loss distribution between the whole combustion chamber and the piston only.

To study the combustion and cycle-resolved heat losses, the apparent rate of heat release dQ_n was calculated with equation (1). A derivation of this expression can be found in the book by Heywood.¹⁰ The following assumptions apply: the cylinder charge is homogeneous with respect to temperature and composition and behaves like an ideal gas with a fixed gas constant R . The ratio of specific heats, κ , is a function of the average charge temperature and composition. The other variables in the equation are the cylinder pressure p , the cylinder volume V , and the crank angle θ

$$\frac{dQ_n}{d\theta} = \frac{\kappa}{\kappa - 1} p \frac{dV}{d\theta} + \frac{1}{\kappa - 1} V \frac{dp}{d\theta} \quad (1)$$

The model from Hohenberg and Killmann¹¹ was used to calculate κ . The cylinder volume was calculated from the engine geometry and crank angle position, and corrected for elastic engine deformation due to cylinder pressure and inertial forces. Using the correct compression ratio for the cylinder volume calculation is very important when comparing heat release rates: an error can be mistaken for an effect of a TBC on the aRoHR. In this investigation, the compression ratio was determined thermodynamically from the motored EOPs.¹² Analysis of the rate of heat release showed that the observed delay of part of the heat release could be properly quantified with the crank angle location of 75% fuel mass fraction burned (MFB75). The definition of MFB75 in this article is the crank angle position where 75% of the total apparent heat release has occurred.

For the calculation of the energy balance, only the high-pressure cycle was considered. According to the first law of thermodynamics, the amount of work $W_{i,g}$, wall heat loss Q_w , and exhaust loss H_{exh} equals the fuel heat input Q_f , according to equation (2). The heat from the fuel was calculated with equation (3), where the energy content of the exhaust emissions is subtracted from the energy content in the fuel

$$Q_f = W_{i,g} + Q_w + H_{exh} \quad (2)$$

$$Q_f = m_f Q_{LHV_f} - m_{CO} Q_{HV_{CO}} - m_{THC} Q_{LHV_{THC}} \quad (3)$$

The indicated work for the high-pressure cycle was calculated with equation (4). The wall heat losses were derived from the difference between the fuel heat and the total apparent heat released Q_n , shown in equation (5). The exhaust enthalpy losses were found by the difference between the total net heat released and the work performed on the piston (equation (6)). The total

net heat released or cumulative apparent heat release was calculated with equation (7) for the closed part of the high-pressure cycle, between -145°CA and 145°CA , with the assumption that the heat transfer before intake valve closing and after exhaust valve opening can be neglected for the high-pressure energy balance calculation

$$W_{i,g} = \int_{-180}^{180} p \frac{dV}{d\theta} d\theta \quad (4)$$

$$Q_w = Q_f - Q_n \quad (5)$$

$$H_{exh} = Q_n - W_{i,g} \quad (6)$$

$$Q_n = \int_{-145}^{145} \frac{dQ_n}{d\theta} d\theta \quad (7)$$

The terms in equation (2) for indicated work, heat transfer, and exhaust enthalpy loss were scaled with the input fuel energy $m_f Q_{LHV_f}$. This results in equation (8) where the ratios of the different contributions then add up to unity. The purpose of the scaling is to make it easier to compare the results within this investigation and with other published results

$$1 = \text{ratWork} + \text{ratHT} + \text{ratEXL} + \text{ratEML} \quad (8)$$

For the motored energy balance, the terms in equation (2) were scaled with the engine swept volume and expressed in terms of mean effective pressure, resulting in equation (9). These terms for indicated work, wall heat transfer, and exhaust loss add-up to zero as there was no input of fuel energy for the motored case

$$0 = \text{IMEPH} + \text{HTMEP} + \text{EXLMEP} \quad (9)$$

To complement the calculated heat loss from the thermodynamic assessment, the heat flow to the piston cooling oil was calculated from the temperature increase in the oil in the cooling gallery and the piston cooling oil flow, according to equation (10)

$$\dot{Q}_{oil} = \rho_{oil} c_{p,oil} \dot{V}_{oiljet} (T_{oilexit} - T_{oiljet}) \quad (10)$$

Also, this energy loss was scaled and defined as ratQOil for fired operation and QOilMEP for motored operation. More details for the actual measurement are found in section "Experimental setup."

Statistical analysis—data modeling with multiple linear regression

Multiple linear regression (MLR) is a statistical method that fits a linear equation to a result (response) from multiple inputs (factors). The input factors can be continuous or categorical with multiple levels. Equation (11) shows the general model equation. The factors are scaled by their range and centered by their median as in equation (12), to be able to compare the relative contribution of each factor. It is possible to combine

factors to capture interactions, but this was not needed in this investigation

$$\text{response} = C_0 + \sum_{i=1}^n (C_{cat})_i + \sum_{j=1}^m ((C_{var})_j \times \text{factor}_j^*) \quad (11)$$

$$\text{factor}^* = 2 \times \frac{\text{factor} - \text{factor}_{median}}{\text{factor}_{max} - \text{factor}_{min}} \quad (12)$$

A number of benefits come from modeling the measured data:

1. Combining data from multiple measurements in each EOP makes the estimation of the results more precise.
2. Fitting a model to the data creates correlation coefficients for the different input factors. With these coefficients, the contribution of each input factor can be studied separately.
3. With the model, it is possible to make predictions and study new combinations of the input factors.
4. The model provides 95% confidence intervals for the calculated results and for the model coefficients.

In this study, MLR models were created for the measurement results from the fired and motored energy balance, the combustion delay, and the exhaust emissions. The input factors for the models were piston surface coating, surface roughness, compression ratio, and factors that represent the experimental conditions. More details about the models and the interpretation of the model coefficients are found in the “Experimental results” section and in Appendix 3. The software used for the statistical analysis was MODDE 11 from Umetrics.

Experimental setup

This section describes the test facility and details about the tested pistons with respect to coatings and surface roughness. The chosen EOPs and the test sequence are explained at the end.

Experimental test facility

The engine used in this experimental investigation was a single-cylinder light-duty diesel engine. The specifications of the engine and the combustion system are listed in Table 1. The cylinderhead and fuel injection system were taken from a four-cylinder Volvo production engine and modified for single-cylinder operation.

The measurement system consisted of a high-speed and low-speed data acquisition system. The high-speed measurements included crank angle resolved

Table 1. Engine and fuel specifications.

Test engine type	AVL 5812
Displaced volume	0.492 dm ³
Stroke	93.2 mm
Bore	82.0 mm
Compression ratio (nominal)	15.5
Bowl type	Re-entrant
Number of valves	Four
Swirl Number (Honeycomb)	2.0–3.2
Nozzle hole number and diameter	8 × 0.125 mm
Included spray angle	155°
Fuel injection system	Common Rail, 2500 bar
Injector actuator type	Solenoid
Fuel	Diesel CN 51, 10% FAME

CN: cetane number; FAME: fatty acid methyl esters (biodiesel).

measurements of cylinder pressure, fuel injection pressure, and injector current. The crank angle resolution was 0.1 °CA during fuel injection and combustion (–90 to 120 °CA BTDC) and 1.0 °CA outside of this window. A selection of calculation results from the real-time cylinder pressure analysis was sampled and stored by the low-speed measurement system as well. The high-speed and low-speed data acquisition started at the same trigger. The high-speed data were recorded for 100 engine cycles. The low-speed data were sampled for 2 min with a frequency of 10 Hz.

The heat transfer to the piston cooling oil was calculated from the temperature increase in the oil in the cooling gallery and the oil flow, according to equation (10). To improve the accuracy of the piston exit oil temperature measurement, a short pipe was added to the cooling oil gallery exit of the piston, directing the oil flow to the temperature sensor. The sensor itself was mounted in a funnel that collected the oil below the piston cooling oil exit (adopted from Dahlström et al.¹³).

Together with the above, the exhaust emissions of NO_x, Soot, CO, and total hydrocarbons (THCs), as well as concentrations of O₂ and CO₂, were measured. The level of CO₂ of the intake charge was measured as well to calculate and control the exhaust gas recirculation (EGR) level. Table 2 shows the most important measured variables and the specification of the sensors used.

Test object details

The two TBC materials selected for this investigation were alumina and zirconia. The alumina TBC was grown from the piston surface by an electrochemical anodizing process, with a target thickness of 200 μm. As the volume of the oxide is twice that of the aluminum, 0.10 mm was machined off the surface where the coating was formed. Typical values for thermal conductivity and heat capacity for this type of coating are 0.7–1.3 W/m K and 860–890 J/kg K.¹⁴

The zirconia TBC with the same target thickness of 200 μm was applied by plasma-spraying after

Table 2. Main measured variables with instrumentation. The fuel pressure sensor is integrated in the Denso injector.

Variable	Sensor/device
Cylinder pressure	AVL GH14P
Crank angle position	AVL 365C
Intake temperature	Pentronic PT100
Intake pressure	GEMS 4000 0–6 bar abs.
Exhaust pressure	GEMS 4000 0–10 bar abs.
Fuel pressure	Denso piezoresistive
Fuel consumption	AVL 733 fuel balance
Piston cooling jet oil flow	Contoil VZF 20 RC 130/16
Piston cooling oil temperatures	Pentronic PT100
Emissions, EGR	Horiba MEXA-7100DEGR
Soot emissions	AVL 415SE Smoke Meter

Table 3. Specifications for surface treatments of the pistons. The roughness R_a is the average value for the whole piston top surface.

Piston ID	Surface treatment	Applied surface	R_a (μm)
Ref	Standard machined	—	1.3
RefRgh	Sandblasted	1, 2, 3, 4	3.4
RefPol	Polished	1, 2, 3, 4	0.2
AlORgh	Anodized	1, 2, 3, 4	7.2
AlOPol	Anodized, polished	1, 2, 3, 4	5.5
AlOTopPol	Anodized, polished	1, 4	3.2
ZrORgh	Plasma-sprayed	1, 2	7.3
ZrOPol	Plasma-sprayed, polished	1, 2	3.0
ZrOTopPol	Plasma-sprayed, polished	1	2.3

1 = squish surface; 2 = bowl bottom; 3 = bowl side; 4 = top land.

sandblasting the surface and application of a 150- μm -thick metallic bond coat. The bond coat and top coat materials were acquired from Oerlikon, with the specifications Amdry 365-2 and Metco 204b-ns, respectively. Typical values for thermal conductivity and heat capacity for this coating are 0.8–1.3 W/m K and 440–460 J/kg K, respectively. To stay close to the reference piston surface profile, 0.35 mm was machined off where the coating was applied. A typical variation of the coating thickness of $\pm 40 \mu\text{m}$ was found for both TBCs using scanning electron microscope (SEM) analysis of the cross section of cut pistons.

The six pistons with TBCs were coated on different locations of the piston top surface (Table 3). The pistons labeled “ZrORgh” and “ZrOPol” were spray coated from above, leaving the vertical bowl wall and the ring top land uncoated. However, the vertical bowl wall was sandblasted and had a rough surface. The piston labeled “ZrOTop” was coated only on the squish surface. The pistons with label “AlORgh” and “AlOPol” were coated on the whole piston top surface, including the ring top land. The piston labeled with “AlOTop” was coated on the squish surface and ring top land only.

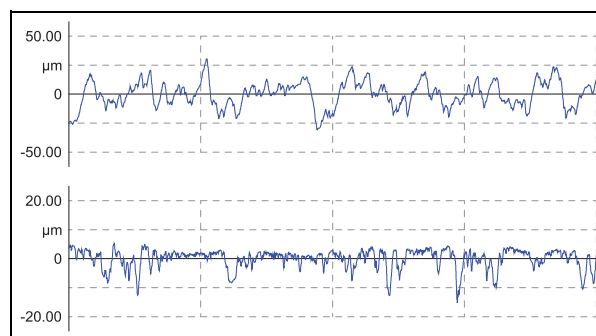


Figure 1. Surface profiles for the rough (top) and the polished (bottom) zirconia surfaces. Measurement length = 4 mm. Note the different scales for the y-axis.

The purpose of the differentiation of the pistons with the top coating was to evaluate the findings of Kawaguchi et al.⁹ that the TBC applied in the bowl had an adverse effect on the combustion process, whereas the coating on the top reduced heat losses and improved indicated efficiency. The difference in the coated surface area between the alumina and zirconia coated pistons was related to the limitations in the used plasma-spraying process.

The surface roughness R_a of the pistons was measured according to ISO 4287. Where the surface roughness varied for different surfaces of the piston top, the measured values were multiplied by their respective surface area, summed and divided by the total surface area, see Table 3. Use of this method assumes that the effect of the surface roughness is proportional with the surface area and does not take local differences in heat transfer rate into account, for example, piston bowl versus squish area.

To investigate the effect of the level of surface roughness, the TBCs on the top-coated pistons and two pistons with coating in the bowl and on the top were polished with fine-grained sandpaper. As can be seen in Figure 1, polishing of the coatings removed the peaks from the surface. However, both TBCs have a porous structure, hence there will remain small holes and cavities in the surface. To be able to assess the effect of surface roughness separately from the TBCs, a mirror polished and a sandblasted uncoated piston were included in the test as well. With the unpolished coated pistons, a total of nine different pistons were tested, as listed in Table 3.

Table 4 lists the piston properties and the order in which the pistons were tested. Figure 2 shows pictures of six of the nine pistons to give an impression of their main features: surface roughness, coating type, and coating location.

Engine Operating Points and test run setup

Three EOPs were chosen to represent low-, medium-, and high-load conditions. Table 5 lists the engine speed, fuel mass, fuel pressure, and injection pressure as well

Table 4. Compression ratio and test run order for the different engine builds.

Run ID	Piston description	CR (-)	Run order
Ref1	Reference	15.48	2
Ref2	Reference	15.48	8
Ref3	Reference	15.52	11
RefRgh	Ref. rough	15.49	3
RefPol	Ref. polished	15.50	5
AlORgh	Alumina original	14.91	1
AlOPol	Alumina polished	15.09	6
AlOTopPol	Alumina top-polished	15.14	10
ZrORgh	Zirconia original	14.90	4
ZrOPol	Zirconia polished	15.08	7
ZrOTopPol	Zirconia top-polished	15.33	9

CR: compression ratio. The CR was thermodynamically determined from motored EOP.¹²

as intake and exhaust pressure. All load points were run with one pilot fuel injection of 2 mg.

In each of the three engine load points, a few parameter variations were included. The parameters were swirl level, fuel injection pressure, and EGR level because they have an effect on heat losses. However, in this investigation, the variation of swirl was excluded from the evaluation. A second purpose of using EGR was to create an increasing amount of soot deposits on the combustion chamber walls during a test run. These soot deposits have an insulating effect and this will be compared with the insulating effect of the TBCs.

Each hardware setup was measured in 26 consecutive EOPs. The test run started with motoring conditions for EOP A, B and C. Thereafter the high-load point C was run with a variation of fuel pressure, followed by motoring. The medium-load point B was run with a variation of EGR and swirl, again followed by motoring. Finally, the low-load point was run, also with a variation of EGR and swirl and concluded with motoring. The purpose of the motored operating points was to calculate the compression ratio and to evaluate the motored heat losses with TBCs and different soot deposit levels. The test run was automated with fixed times for the stabilizing of boundary conditions and the recording of measurements. A detailed description of the complete test run setup can be found in Table 9 in Appendix 2.

Experimental results

In this section, results from the measurements and data modeling are presented. The complete data models, represented by their coefficients in bar graphs, can be found in Appendix 3. All modeled results for the energy balance, combustion phasing, and emissions are found in Appendix 4.

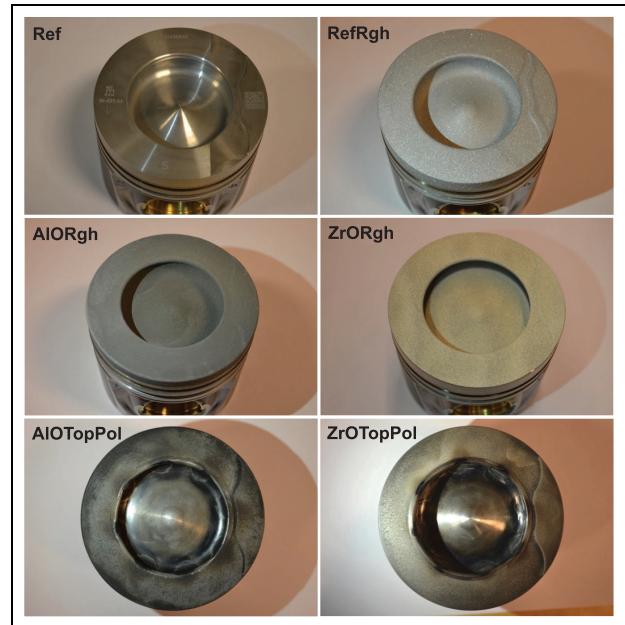


Figure 2. A selection of the tested pistons. See Tables 3 and 4 for details.

Indicated efficiency and piston heat loss

In Figure 3, the heat loss to the piston cooling oil is plotted. All 26 measurement points of the test runs are included here. The data points with the low heat loss are the motored points before, between and after the fired EOP. A rise in fuel pressure increases heat losses (EOP C) and increasing EGR reduces the heat losses in EOP B and C, as expected. The highest heat loss occurs for the uncoated rough piston, and the lowest heat loss is seen for the coated pistons. The polished piston variants have lower heat losses than the rough variants, except for piston AlORgh. The low heat loss for this piston was probably caused by slower combustion.

In Figure 4, the measured indicated efficiency is presented for a selection of the fired run numbers in the test sequence: run numbers with swirl variation and fuel cut-off were excluded. The variation of efficiency level is caused by an increase in fuel pressure for EOP C and an increasing EGR rate for EOP B and EOP A. It can be clearly seen that the polished pistons show a higher efficiency than the rough pistons, which is in line with the lower heat losses seen in Figure 3. But the coated pistons exhibit a lower efficiency than the uncoated pistons despite the lower heat losses to the piston cooling oil.

The first step for further analysis of this discrepancy and of the effect of surface roughness is to compare the aRoHR for the different pistons and coatings.

Heat release rate and combustion phasing

The aRoHR, calculated from the cylinder pressure, shows any addition or removal of energy to the

Table 5. Base settings for the low-, medium-, and high-load engine operating points. For the complete test run table see Table 9 in Appendix 2.

EOP label	Speed (r/min)	m_fuel (mg/str)	p_fuel (bar)	p_intake (bar)	p_exhaust (bar)
A	1500	2 + 13	500	1.2	1.4
B	1500	2 + 28	1000	1.7	1.9
C	3000	2 + 58	2000	2.6	2.8

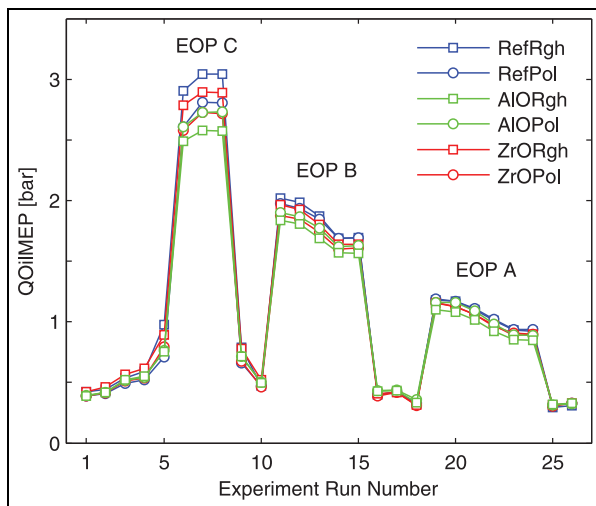


Figure 3. Heat energy to piston cooling oil versus RunNr, according to Table 9 in Appendix 2. The low values for the heat loss to the piston cooling oil are from the motored EOP.

cylinder charge which is not related to the piston work. It reveals the heat released by combustion and the heat loss to the walls as well as enthalpy loss from blow-by or loss of heat for fuel evaporation. Heat release from combustion and heat losses occur at the same time during combustion, and it is not possible to distinguish between them. This has to be kept in mind when using the aRoHR for investigating the effect of TBC and surface roughness on combustion and heat losses.

In Figure 5, the aRoHR is plotted for the pistons with high surface roughness and the reference piston, for all three load cases. The numbers 1–3 in the figure indicate the following: first wall–jet interaction, end of injection (EOI), and pilot–main interaction, respectively. For the high-load case EOP C, there is an obvious reduction of the peak aRoHR for the rough and coated pistons. Mostly after start of wall–jet interaction, until a few CA degrees after EOI. After 20°CA, the aRoHR is higher for the rough pistons compared to the reference piston, which indicates late burning or reduced heat losses early in the expansion stroke.

The same effect as for EOP C can be seen for the medium-load EOP B for the rough coated pistons during the wall–jet interaction. The aRoHR from the rough uncoated piston remains close to the reference piston in this EOP. In the low-load operation point A,

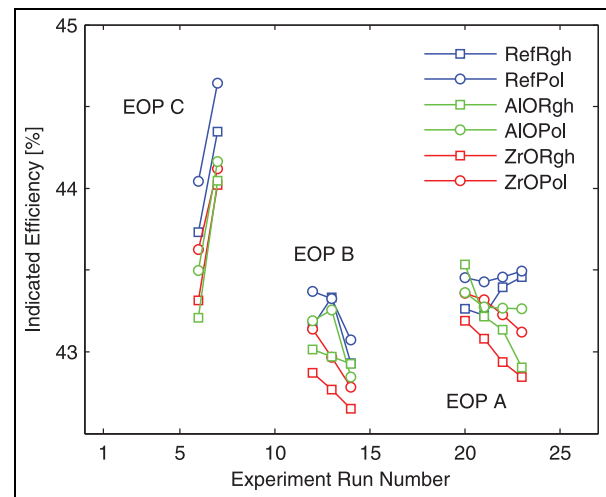


Figure 4. Gross indicated efficiency versus RunNr, according to Table 9 in Appendix 2. EOP C is measured with different fuel pressures, EOP B and A show an increasing EGR level.

the injection ends before wall–jet interaction occurs. For the low-load case, there seems to be an interaction between the main and the pilot combustion. The heat release from the main fuel injection is shifted earlier when the pilot ignites later. The ignition delay of the pilot fuel seems to be correlated with the compression ratio. It is not obvious if there is an effect from the surface roughness on the aRoHR. However, the aRoHR in the tail of the combustion for the rough AIO piston is slightly raised compared to the other pistons.

For the polished pistons and the reference piston, the curves for the aRoHR are shown in Figure 6. The polished uncoated piston now has the highest aRoHR during wall–jet interaction for EOP C. The polished coated pistons still have an aRoHR lower than the reference piston, but the difference is significantly smaller. Even here, there is a late combustion or reduced heat loss in the tail of the combustion. In EOP B, the difference between the standard and polished piston cannot be distinguished. The coated pistons still show a somewhat lower aRoHR relative to the reference piston. The interaction between pilot and main injection in EOP A is also smaller for the polished pistons, probably because they have a compression ratio (CR) closer to the reference piston and a more similar ignition delay.

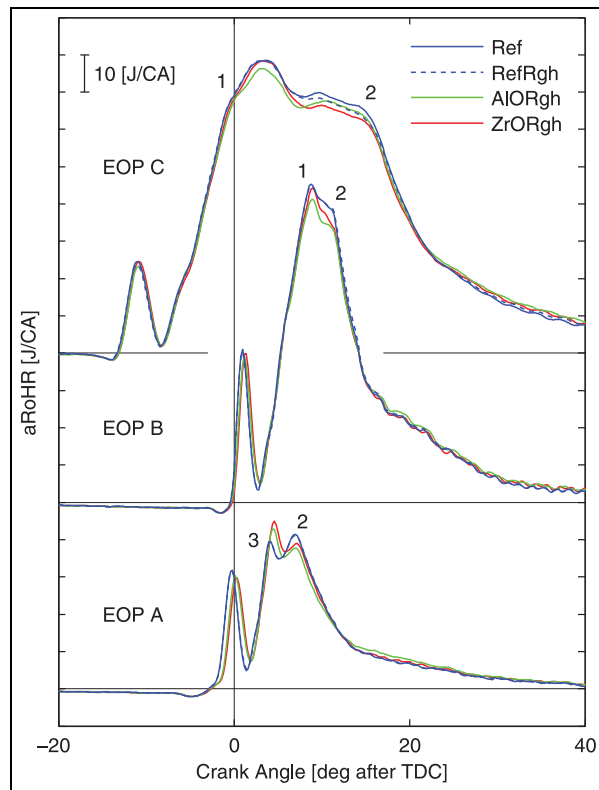


Figure 5. Apparent rate of heat release for the rough pistons for run numbers 7, 12, and 20 in Table 9 in Appendix 2 (1 = start spray-wall interaction; 2 = end of fuel injection; 3 = pilot-main interaction).

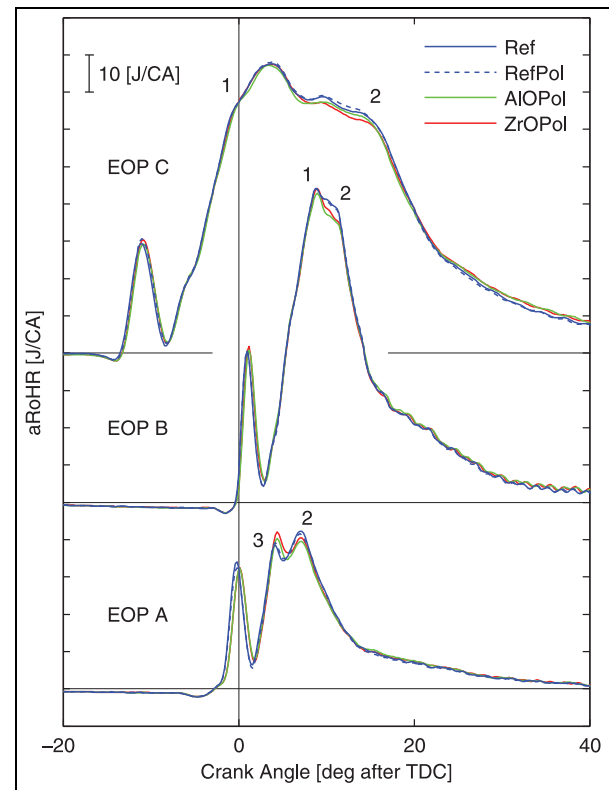


Figure 6. Apparent rate of heat release for the polished pistons for run numbers 7, 12, and 20 in Table 9 in Appendix 2 (1 = start spray-wall interaction; 2 = end of fuel injection; 3 = pilot-main interaction).

When comparing the top-coated polished pistons with the reference piston in Figure 7, an increase in the aRoHR just before EOI becomes evident for EOP C. The piston has traveled downward to a position where the combusting jet targets the top of the piston. An increase in the aRoHR at this point indicates a reduction of heat losses caused by the presence of a coating on the piston top. This effect is not seen for EOP B, probably because the fuel injection ends too early. A small reduction in aRoHR can be noticed here for the top-coated pistons. For the lowest load point, EOP A, the aRoHR curves have become more similar again as the compression ratios for the top-coated pistons are closer to the reference piston.

To summarize, a high surface roughness decreased the peak aRoHR for the medium- and high-load EOPs when there is an interaction between the burning jet and the piston wall. For the high-load point, there is also a significant increase in the aRoHR in the tail of the combustion. There are two mechanisms that can cause these changes: (1) a longer combustion duration and (2) an increased heat loss during wall-jet interaction, followed by a reduced heat loss early in the expansion stroke.

Data modeling results

MLR was used to create models from the measured data as described in the “Method” section. Each

response parameter was modeled separately, for each EOP. The number of observations used for the models was 22 for EOP C, 33 for EOP B, and 44 for EOP A.

Model coefficients close to zero were excluded from the models to improve model quality. Coefficients with very low confidence were removed as well as they did not contain useful information. The coefficients for all models can be found in Appendix 3. A selection of the model coefficients and responses will be presented and discussed here.

In Table 7, the input factors for the models are listed with their levels or their variable range. The compression ratio and the surface roughness are continuous, qualitative input factors. The coating type, fuel pressure, EGR level, and soot deposit level are discrete, categorical input factors. The variation of fuel pressure and EGR was included in the models to be able to combine the measurements for each EOP in one model. The compression ratio was included because of the relatively large range and its effect on the energy balance.

Early in the test run of the polished uncoated piston (Run order 5), a leakage of the injector washer was detected. This leakage exposed the injector to high-temperature gas which changed the pilot injection slightly. This change of the injector behavior was modeled by the factor AB: “A” represents the injector before the leakage, “B” after the leakage. The factor

Table 6. Effect of coating type, coated surface (‘ indicates top-coated), surface roughness and compression ratio on the motored and fired energy balance, the combustion speed, and the exhaust emissions, relative to the reference piston. A positive correlation is indicated by an upward arrow. Summary of the figures in Appendix 3.

	EOP C / high load						EOP B / medium load						EOP A / low load					
	AIO	ZrO	AIO'	ZrO'	R _a	CR	AIO	ZrO	AIO'	ZrO'	R _a	CR	AIO	ZrO	AIO'	ZrO'	R _a	CR
IMEPH	↓	↓	↑	↑	∧	—	↓	↓	—	—	—	—	∨	∨
HTMEP	↓	↓	∨	—	↑	↑	—	—	—	—	∧	∧
QOiiMEP	↓	∨	—	—	↑	↑	∨	∨	∨	∨	.	.	—	—	—	—	∧	—
EXLMEP	∧	∧	—	—	↓	↓	—	—	—	—	∨	∨
ratWork	—	—	—	—	↓	—	↑	↓	—	—	↓	—	↑	—	↓	∨	↓	—
ratHT	↓	↓	↓	—	↑	.	↓	∨	↓	∨	↑	↑	∧	↑	∧	—	↑	↑
ratQOil	↓	↓	↓	∨	↑	.	↓	↓	↓	↓	↑	↑	∧	↑	∧	—	↑	↑
ratEXL	↑	↑	↑	—	—	.	↑	↑	↑	∧	↓	↓	∨	↓	∨	—	↓	↓
MFB75	—	—	—	—	↑	—	—	—	—	—	—	↓	↓	↓	—	—	—	↓
EINOx	↑	↑	↑	∧	↓	∧	—	—	—	—	∨	—	∧	—	—	—	—	↑
EISoot	—	—	—	∧	—	—
EITHC	∨	∨	—	.	∧	—	↓	↓	—	.	↑	—	↑	↑	↑	.	—	—
EICO	↓	∨	—	∨	↑	—	↓	↓	—	∨	∧	—	↓	∨	—	—	—	↓

↑↓ represents significant with a 95% confidence level; ↑↓ represents clear trend; ∧∨ represents weak trend; — represents neutral; . represents low confidence data.

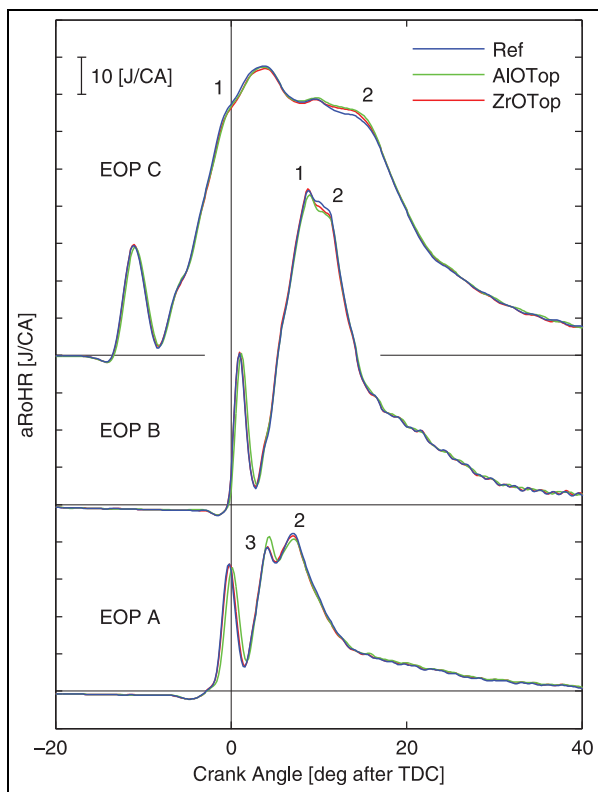


Figure 7. Apparent rate of heat release for the top-coated and polished pistons for run numbers 7, 12, and 20 in Table 9 in Appendix 2 (1 = start spray-wall interaction; 2 = end of fuel injection; 3 = pilot-main interaction).

AB was significant for the exhaust emission models and the crank angle position for MFB75. The energy balance for fired and motored engine operation was not affected.

The soot deposit level increased during the test run for each fired EOP, especially with the application of EGR. The effect of soot deposits on the energy balance and emissions is effectively integrated in the categorical factor EGR. Consequently, this factor not only shows the effect of the intake CO₂ level but also incorporates the effect of increasing soot deposits.

Below, an example is given of the way to calculate a response; in this case, the indicated efficiency (ratWork), for the standard reference piston, run number 12 in Table 9 in Appendix 2, EOP B. The values for the coefficients are taken from Figure 9. Centering and scaling is applied for the compression ratio and surface roughness factors: CR = 15.5 and R_a = 1.3 μm become CR* = 1.0, R_a* = -0.7

$$\begin{aligned}
 \text{ratWork} &= C_0 + C_{TBC(Ref)} + C_{AB(A)} \\
 &\quad + C_{EGR(E0)} + C_{CR} \times CR^* + C_{R_a} \times R_a^* \\
 \text{ratWork} &= 43.09 + 0.005 + 0 \\
 &\quad + 0.025 + 0.008 \times 1.0 + -0.30 \times -0.70 \\
 &= 43.34\%
 \end{aligned}$$

From this example, it can be seen how the factor coefficients express the correlation between the input factors and the responses. These model coefficients illustrate the magnitude and confidence level of the correlations between input (factors) and output (responses) of the experimental data.

Table 6 gives an overview of the modeled correlations between the factors and responses for low-, medium-, and high-load engine operation. Note that a high confidence level of a correlation coefficient does not necessarily mean a large effect of the factor on the response. In Appendix 3, all modeled responses are found for the three EOPs. The figures show results for

Table 7. The input factors for the regression models.

Factor	Type	Category name or variable range
TBC	Categorical	Ref, AlO, ZrO, AlOTop, ZrOTop, or: Ref, AlO, ZrO AlO', ZrO' or: Rf, Al, Zr, Al', Zr' – 1 to 1 (14.9 to 15.5) – 1 to 1 (0.2 to 7.3 μm)
CR	Variable	P1, P2
R_c	Variable	E0, E1, E2, E3
p_{fuel}	Categorical	S0, S1, S2, S3 (motored only)
EGR	Categorical	Run order 1–4 \Rightarrow A; 5–11 \Rightarrow B
Soot	Categorical	
AB	Categorical	

TBC: thermal barrier coating; CR: compression ratio; EGR: exhaust gas recirculation. The labels in the TBC category are abbreviated where space in the figures is limited. See also Table 3 for details.

indicated efficiency, heat losses, exhaust losses, combustion phasing, and emissions as well as the indicated mean effective pressure for the motored cases.

Indicated work, heat loss, and exhaust loss

Figure 8 shows the measured indicated efficiency for the reference piston, the fully coated rough pistons, and the polished top coated pistons. The estimations for the measured indicated efficiency from the model are very precise. The 95% confidence interval is close to ± 0.1 percentage point. This shows that the engine operating conditions and measurements were repeatable and that the model fits the data well. The high precision is valid when comparing the estimation of measured results within this group of experiments.

The bar graph shows the same trend as Figure 4. The fully coated rough pistons (AlORgh and ZrORgh) have a significantly lower indicated efficiency compared to the reference piston. The top-coated pistons AlOTop and ZrOTop show a tendency to a lower efficiency compared to the reference piston, but the difference is very small. To investigate the cause for the lower efficiency of the fully coated rough pistons, the coefficients of the MLR models for the energy balance can be used.

Figure 9 shows the model coefficients for indicated efficiency, wall heat loss, and heat loss to the piston cooling oil for the medium-load point B. This figure is used as an example, selected from the figures in Appendix 3, where the coefficients for all MLR models can be found. The confidence intervals for the coating type and compression ratio are relatively large because these two factors are correlated. However, the coefficients for coating type and compression ratio show a very consistent behavior in the different models despite their collinearity; therefore, the models are still considered useful for analysis.

The coefficients for the coatings indicate a reduction in heat losses of around 0.5 percentage point compared to the reference piston. The pistons with the alumina

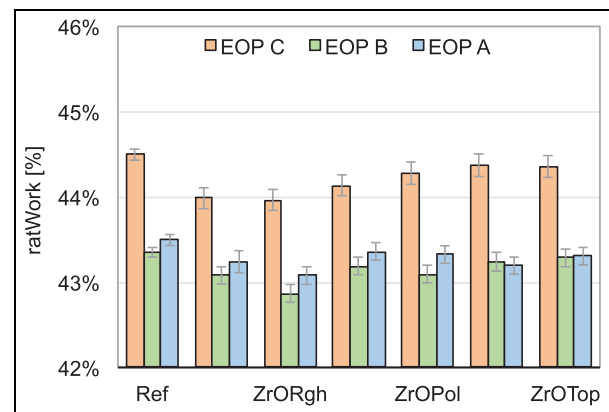


Figure 8. Estimation of the measured indicated efficiency from the modeled data. No EGR, high rail pressure for EOP C, representing run numbers 7, 12, and 20 in Table 9 in Appendix 2. The error-bars represent 95% CI.

coating seem the most effective in heat loss reduction. The effect of the coatings on the indicated efficiency is small, in the order of ± 0.1 percentage point. An increase in the compression ratio increases the heat losses but has no significant effect on the indicated efficiency. The EGR level is the most significant factor for the heat losses. Increased EGR levels result in lower heat losses. The indicated efficiency does not increase with increased EGR because the combustion is slower, which can be seen in Figure 17, EOP B, chart (e) in Appendix 3.

The surface roughness shows a rather strong and significant correlation with all three responses of the energy balance. Higher surface roughness increases heat losses and reduces indicated efficiency. The model coefficients for the heat loss to the piston cooling oil and the calculated heat loss from the cylinder pressure data show a remarkable agreement, keeping in mind that these measurements are completely independent. The difference in coefficient magnitude can be explained by the fact that the heat loss to the piston cooling oil is just a part of the total heat loss.

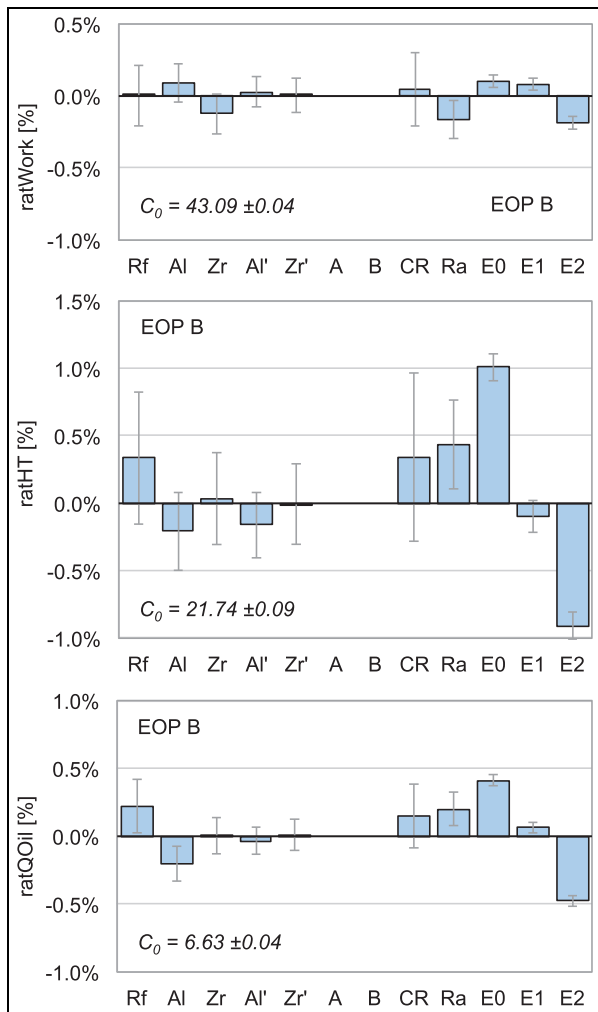


Figure 9. Model coefficients for indicated efficiency, wall heat losses, and heat losses to the piston cooling oil for EOP B. The error-bars show the 95% CI. The legend for the coefficient labels can be found in Table 7.

To give an overview of the effect of surface roughness on the energy balance for all three EOP, Table 8 shows how a reduction of surface roughness R_a from 7.3 to 0.2 μm changes the indicated efficiency, heat losses, and exhaust losses. The indicated efficiency is increased with 0.9% for EOP B and C, and 0.7% for EOP A. Reduced heat losses are the main cause for the efficiency improvement. For the high-load EOP C, MFB75 was reduced as well. Exhaust losses increased as expected when there is a reduction of heat losses. The emission loss (ratEML) was very small and not shown in the table. This loss can be found implicitly as the remaining term to get 100% when adding up the terms for ratWork, ratHT, and ratEXL.

To visualize the effect of smooth coatings with the same surface roughness and compression ratio as the reference piston, the model was used to predict the indicated efficiency for all three load points, shown in Figure 10. The (*) with the coating category labels indicates that the coating properties are predicted. There is no significant difference between the coatings and the

Table 8. Predicted effect of a roughness (R_a) decrease of the piston surface from 7.3 to 0.2 μm on the energy balance and MFB75.

EOP label	ratWork (%)	ratHT (%)	ratEXL (%)	MFB75 ($^{\circ}\text{CA}$)
A	+0.3	-0.6	+0.3	
B	+0.4	-0.8	+0.4	
C	+0.4	-0.5	+0.1	-1.4

MFB75 is not significantly changed for EOP A and B. For 95% confidence intervals, see Figure 17 in Appendix 3.

reference piston anymore. Predictions for all terms in the energy balance are found in Appendix 4.

To summarize, the low indicated efficiency for the unpolished coated pistons is caused by their high surface roughness. High surface roughness increases heat losses to the combustion chamber walls and delays MFB75 for the high engine load point C. The effect of the coatings on the indicated efficiency is small, in the order of ± 0.1 percentage point.

Exhaust gas emissions

The effects of the TBCs and surface roughness on the exhaust emissions are relatively small but significant. Figure 11, selected from Appendix 3, shows the correlation coefficients in the model for the high-load point for the fuel-specific NO_x , Soot, and CO emissions.

The formation of NO_x emissions is sensitive for the maximum local temperatures in the combustion chamber. A rise in NO_x emission indicates an increase in the maximum local temperatures. It is, therefore, expected that NO_x emissions increase when TBCs are applied, see Figure 11. That a higher surface roughness reduces NO_x emissions is logical due to the higher heat losses and due to the slower combustion for the high-load point. The presence of an alumina coating with a median R_a level on the whole piston top increases NO_x about 8%. A surface roughness R_a of 7.3 compared to 0.2 μm reduces NO_x approximately 12% for the high load. Furthermore, the effect of a higher compression ratio and higher rail pressure is also in line with the expectations.

The soot emissions are less predictable compared to NO_x . This is because both soot formation and soot oxidation are temperature dependant. The soot levels in the high-load point are generally very low and the effect of the coatings and surface roughness seem to be small as well. The large error-bars indicate that there is no significant trend from the coatings and surface roughness.

The oxidation and reduction of CO is enhanced by high temperature and mixing. It is, therefore, expected that this emission is reduced by the presence of the full coatings and increased by a high surface roughness as shown in Figure 11. There is also a significant effect from the different pilot injection quantities represented

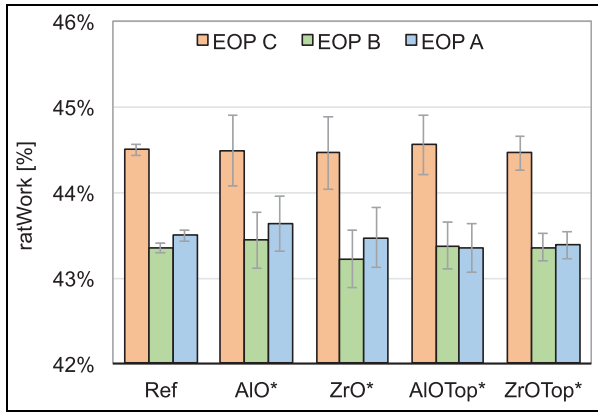


Figure 10. Predicted indicated efficiency (ratWork) with $R_a = 1.3 \mu\text{m}$, CR = 15.5 for all pistons. No EGR, high rail pressure for EOP C, representing run numbers 7, 12, and 20 in Table 9 in Appendix 2. The error-bars show 95% CI.

by the factors A and B. The smaller pilot quantity (A) shows less CO emissions. The levels of CO emissions in this high-load point are low, and the changes from the presence of the coatings and surface roughness are very small.

For the low-load points, where EGR is used, the relative impact of the coatings and surface roughness becomes very small, even though the trends are the same as for the high-load point as illustrated in Figure 11.

The THC emissions show an interesting difference in behavior for the low-load point compared to the medium- and high-load points, see Figure 12. As for CO, the oxidation and reduction of THC is supported by high temperature and mixing. The level of THC is reduced significantly by the fully coated pistons and increased considerably with high surface roughness. However, the low-load point shows the opposite effect. The presence of a coating increases the THC emissions while the surface roughness influence is insignificant.

There might be two competing mechanisms that could explain this observation. A high charge temperature will promote oxidation of unburned fuel. However, a TBC with open porosity might absorb part of the near wall fuel which would prevent the oxidation of this fuel. At a lower engine load and lower charge temperature, the effect of the porosity might dominate, resulting in increased levels of THC emissions. Finally, the effect of the change in injector behavior, modeled by the factor AB can be seen for all emissions. Most likely, this is related to small changes in the pilot fuel quantity.

To summarize, the effect of the coatings and surface roughness on emissions is as expected, with the exception of the THC emissions in the low-load point, where the presence of the TBCs increased THC emissions.

Motored losses

The motored energy balance does not have the complex contribution of combustion on the heat transfer.

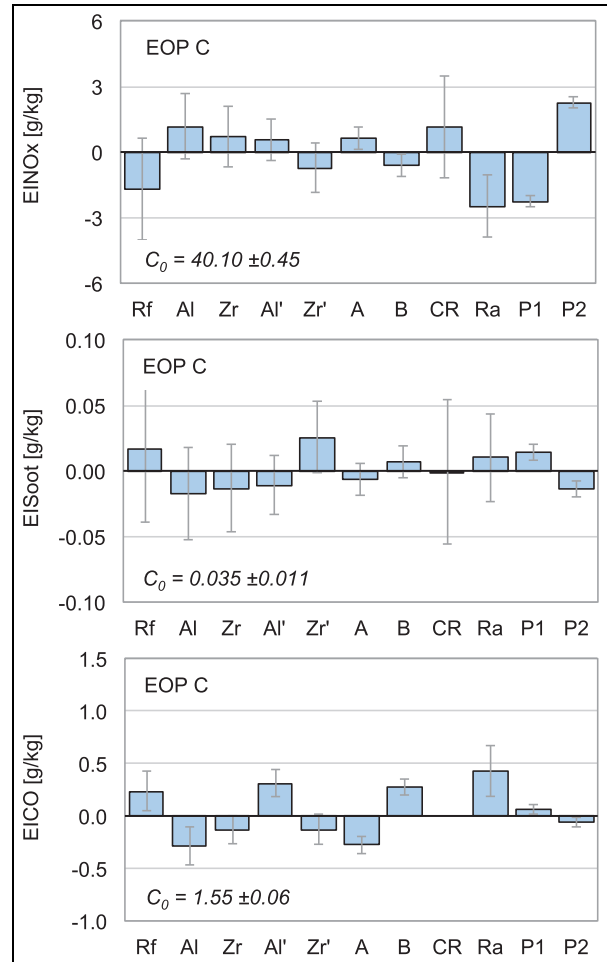


Figure 11. Model coefficients (correlations) for EINO_x, ElSoot, and EICO, for EOP C. The error-bars represent 95% CI. The legend for the coefficient labels can be found in Table 7.

Analysis of the motored energy balance can complete and confirm the findings from the fired energy balance, or shed new light on motored phenomena. Without combustion, the split between indicated work, heat loss, and exhaust loss is fairly constant. Therefore, only the motored losses (IMEPH) will be discussed here. The effect of soot deposits is studied by comparing the motored losses before and after high-, medium-, and low-load engine operations.

The indicated work (IMEPH) for the different pistons and all three EOPs is shown in a bar graph in Figure 13. The major cause for the motored losses in the high-pressure cycle is heat transfer. It is, therefore, expected to see lower motoring losses (a higher IMEPH) for the coated pistons. But this is not the case for the highest load point, where the losses seem to increase for the AlORgh and ZrORgh pistons.

To explain this result, Figure 14 is presented with the model coefficients for EOP B, the medium-load point. The figure shows that the fully coated pistons have a lower motored loss, as expected. Also, the lower compression ratio for these pistons reduces the motored losses. But the high surface roughness increases heat

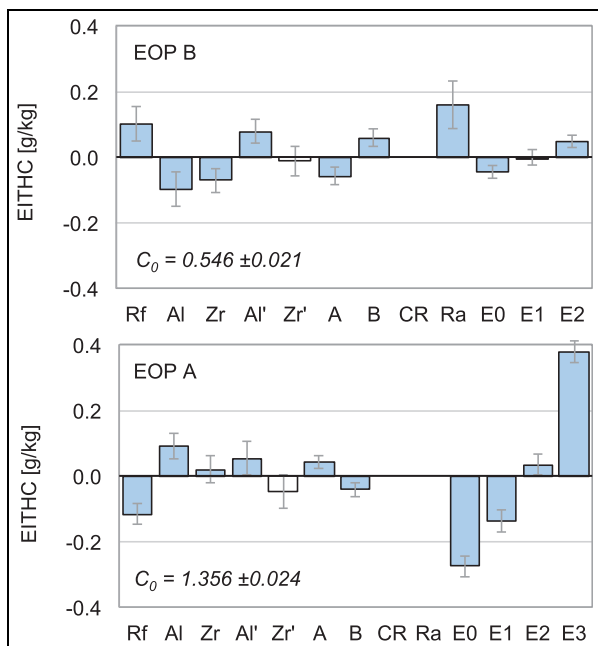


Figure 12. Model coefficients (correlations) for EITHC for EOP A and B. The error-bars represent the 95% CI. The bar for Zr' coefficient is white because there was a negative offset for the THC emissions with the zirconia top-coated piston. The legend for the coefficient labels can be found in Table 7.

losses and counteracts the effect of insulation and compression ratio. The principle effect is the same for all three load cases, which can be seen in Appendix 3. The negative impact of the surface roughness is most significant for the medium and high loads.

Figure 14 also shows the correlation between soot deposits and indicated work, which is much larger than the correlation for the TBCs. Soot appears to be a very good insulator, at least for the motoring conditions.

In Figure 15, a comparison is made between the indicated work for clean conditions and the soot deposit level after motored operation for each load case. The figure shows the results for the reference piston, but the modeled effects of soot deposits are the same for all pistons. The soot deposits reduce the indicated mean effective pressure with about 0.2 bar. This is a significant reduction; typical motored friction losses for a modern LD CI engine, running at 1500–3000 r/min, range from 0.5 to 0.9 bar.¹⁵

To summarize, a high surface roughness counteracts the positive effect of the TBCs on the motored losses. Soot deposits appear to be an effective insulator, which has been shown in the literature,^{16,17} but has not been quantified and presented in this way before. This result also shows the importance of taking into account the effect of soot deposits when investigating TBCs.

Volumetric efficiency

Although the mass flow through the engine was kept constant, small variations were observed for the

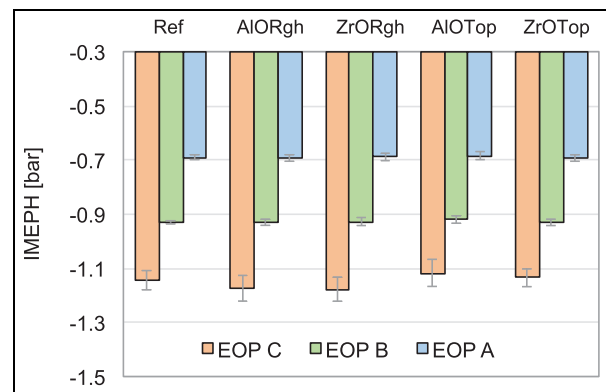


Figure 13. Measured motored losses, unsooted (S0), representing run numbers 1, 3, and 5 in Table 9 in Appendix 2. The error-bars show the 95% CI.

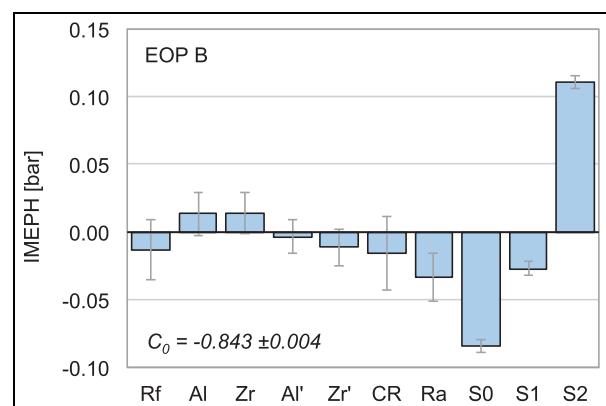


Figure 14. Model coefficients for IMEPH in EOP B. The error-bars represent the 95% CI. The legend for the coefficient labels can be found in Table 7.

volumetric efficiency (0.88–0.90). From the MLR model, it could be seen that a high surface roughness reduced volumetric efficiency more than the presence of a TBC. An explanation for this could be that the high surface roughness increased the heat transfer coefficient. The higher heat transfer would result in an increased surface temperature at the end of the exhaust stroke. During gas exchange, the high heat transfer coefficient combined with the high surface temperature would increase the heat transfer to the fresh charge more than a smooth TBC, resulting in the observed lower volumetric efficiency.

Conclusion

A robust experimental method was designed and applied for the evaluation of two types of TBCs including a variation of surface roughness and coating location. Measurements were made for low-, medium-, and high-load engine operating conditions. Models were fitted to the resulting data using MLR. The use of these models together with robust measurement data made it possible to distinguish small differences in results

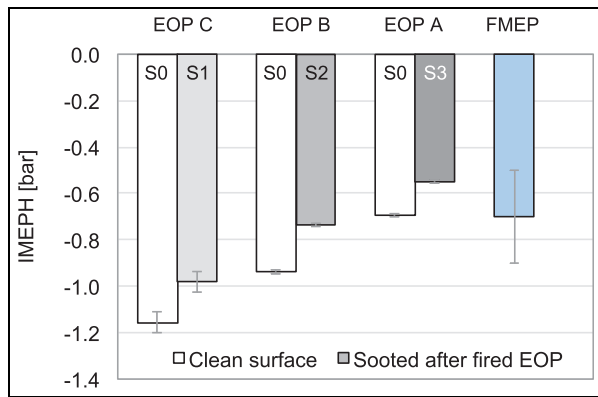


Figure 15. Effect of soot deposits on IMEPH shown for the reference piston. Typical friction losses (FMEP) are included for comparison. The error-bars represent the 95% CI, and the range for FMEP is from 1500 to 3000 r/min.

between the different hardware setups and to make well-supported conclusions regarding the effects of TBC and surface roughness on engine efficiency and heat losses. The key findings from this study are as follows:

- The applied method for the experimental setup and the statistical modeling with MLR resulted in a 95% confidence interval of ± 0.1 percentage point for the indicated efficiency.
- The indicated fuel consumption for the fully coated unpolished zirconia and alumina pistons was up to 1% higher compared to the indicated fuel consumption for the uncoated reference piston.
- Factor analysis revealed that the main cause for the higher fuel consumption of the unpolished coated pistons was their high surface roughness. The high surface roughness increased heat transfer to the walls for all three investigated load points and delayed the combustion in the high-load EOP.
- The pistons with a coating of polished zirconia or alumina, limited to the squish surface, showed no significant difference in fuel consumption compared to the reference piston.
- Analysis of the aRoHR for the different pistons also showed a consistent negative effect of high surface roughness for the medium- and high-load cases. This effect was most visible when the burning jet interacted with the piston surface and in the burn out phase during the expansion stroke.
- The effects of the TBCs and surface roughness on the exhaust emissions were as expected and related to changes in charge temperature. The only unexpected observation was the increase in the THC emissions in the presence of the tested TBCs, for the low-load EOP. This increase might be related to the porosity of the two coatings tested in this investigation.
- Experimental data from motored engine operation showed that soot deposits reduced heat losses more

than the TBCs. The effect of soot deposits should be accounted for when performing experiments with TBCs.

The study was applied on two types of TBC. Other coating types may give different results with respect to engine efficiency and heat losses. However, the robust experimental method and the data modeling using MLR as described in this article should be applicable for many investigations of thermal insulation and TBCs.


Declaration of conflicting interests

The author(s) declare no potential conflicts of interest with respect to the research, authorship, and/or publication of this article.

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Appendix 1

Notation

κ	ratio of specific heats
ρ	density
θ	crank angle
c_p	specific heat capacity
h_c	heat transfer coefficient
H_{exh}	exhaust enthalpy
p	cylinder pressure
Q_f	heat from combusted fuel
Q_n	apparent (net) heat release
Q_w	wall heat loss
\dot{Q}	heat flux
R_a	average surface roughness, ISO 4287
T	temperature
V	cylinder volume
V_s	swept volume
\dot{V}	volume flow
$W_{i,g}$	gross indicated work

Abbreviations

AIO	Aluminum oxide or alumina
aRoHR	Apparent rate of heat release
BSFC	Brake specific fuel consumption
CA	Crank angle
CI	Compression ignition

CI	Confidence interval
CN _r	Condition number of MLR model
CO	Carbon monoxide
CO ₂	Carbon dioxide
CR	Compression ratio
EGR	Exhaust gas recirculation
EI	Emission index
EOI	End of (fuel) injection
EOP	Engine operating point
EXLMEP	Exhaust energy loss
FMEP	Friction mean effective pressure
FSN	Filter soot number
HTMEP	Wall heat transfer energy
IMEPH	Indicated work high pressure cycle
LHV	Lower heating value
MEP	Mean effective pressure
MFB75	CA for 75% mass fraction burned
MLR	Multiple linear regression
NO _x	Nitrogen oxides
Q ₂	Predictive squared correlation coefficient
QOilMEP	Piston cooling oil energy
R ²	Adjusted coefficient of determination
ratWork	Ratio of work to fuel energy
ratHT	Ratio of heat transfer to fuel energy
ratEXL	Ratio of exhaust loss to fuel energy
ratEML	Ratio of energy loss in unburned emissions to fuel energy
TBC	Thermal barrier coating
THC	Total hydrocarbons
YSZ	Yttria stabilized zirconia
ZrO	Zirconium oxide or zirconia

Appendix 2

Table of engine operating points

This test run of 26 engine operating points (EOPs) was designed for the evaluation of combustion chamber heat losses and emissions for thermal barrier coatings (TBCs), including the effect of EOP, fuel pressure, EGR, and swirl. The motored points were dedicated as check points for the compression ratio and for assessment of heat losses with increasing levels of soot deposits. For run numbers 8, 15, and 24, the fuel injection was stopped after 50 of 100 measured cycles during cylinder pressure data acquisition, to investigate the cycle-to-cycle effects of wall temperature transients. For this article, the runs with increased swirl and fuel cut-off were excluded from the evaluation.

All operating points were run in one, automated test sequence for each hardware setup which took 2 h 43 min. The fuel quantity for the pilot and main injection was held constant, resulting in an IMEPH of approximately 5, 10, and 20 bar for the fired engine operating points. Oil and coolant temperatures were set at 90 °C, and the charge intake temperature was set at 50 °C and adjusted to maintain constant air mass flow. Each test run started with non-sooted (first time run or cleaned) parts. The soot numbers for the highest EGR levels were about 2.5 filter soot number (FSN).

Table 9. Definition of engine operating points in the used test sequence. The model label S represents the soot deposit level, P the fuel pressure level, and E the EGR level. Test runs without model labels were not included in the MLR models.

Run number	Model labels	Engine speed (r/min)	Injected fuel (mg/stroke)	Fuel pressure (bar)	EGR (% CO ₂)	High swirl	Intake pressure (bar)	Exhaust pressure (bar)
1	S0	1500					1.2	1.4
2		1500				y	1.2	1.4
3	S0	1500					1.7	1.9
4		1500				y	1.7	1.9
5	S0	3000					2.6	2.8
6	P1	3000	60	1500	0		2.6	2.8
7	P2	3000	60	2000	0		2.6	2.8
8		3000	60/0	2000	0		2.6	2.8
9	S1	3000					2.6	2.8
10	S1	1500					1.7	1.9
11		1500	30	1000	0	y	1.7	1.9
12	E0	1500	30	1000	0		1.7	1.9
13	E1	1500	30	1000	1.5		1.7	1.9
14	E2	1500	30	1000	3.0		1.7	1.9
15		1500	30/0	1000	3.0		1.7	1.9
16	S2	1500					1.7	1.9
17		1500				y	1.7	1.9
18	S2	1500					1.2	1.4
19		1500	15	500	0	y	1.2	1.4
20	E0	1500	15	500	0		1.2	1.4
21	E1	1500	15	500	1.5		1.2	1.4
22	E2	1500	15	500	3.0		1.2	1.4
23	E3	1500	15	500	4.0		1.2	1.4
24		1500	15/0	500	4.0		1.2	1.4
25	S3	1500					1.2	1.4
26		1500				y	1.2	1.4

EGR: exhaust gas recirculation. S0 = cleaned combustion chamber surfaces; S1 = low soot deposits; S2 = medium soot deposits; S3 = high soot deposits.

Appendix 3

Coefficients for the MLR models

The method of multiple linear regression (MLR) fits a linear curve to the experimental data with multiple inputs for one response at a time. The model coefficients for all results from this investigation are presented here, showing the correlation between the different input factors and the modeled output responses. Figure 16 in Appendix 3 shows the model coefficients for the terms of the energy balance for the motored engine operating points. The model coefficients for fired operation are found in Figures 17 and 18 in Appendix 3, showing the terms of the energy balance and engine out emissions, respectively.

The models for the high-load EOP C are based on the measurements from run numbers 6 and 7 (Appendix 2), with a variation of fuel pressure for 11 different hardware builds, which means a total of 22 data points. The models for the medium-load point B are based on the measurements from run numbers 12, 13, and 14, with varying EGR levels for 11 different hardware builds, which means a total of 33 data points. The models for the low-load point C are based on the measurements from run numbers 21, 22, 23, and 24, with varying EGR levels for 11 different hardware builds, which gives a total of 44 data points.

The response models were created and validated using MODDE, a statistical tool for data analysis from Umetrics. Some of the data of outliers were excluded, and in some cases, some of the input factors were excluded to improve model quality. The bar charts include measures for the quality of the model. The adjusted R² value shows how well the model explains the variation within the data with respect to the number of variables. A value of 0.8 or higher is considered good for R². The Q² value is a measure of the predictive capability of the model. This value should be 0.5 or higher to give valuable predictions. The condition number CNr of the model indicates how well separated or independent the input factors are. A value of 10 or lower means good separation. A value of 15 or higher indicates collinearity, a correlation between two or more input factors. For many of the presented models, the condition number is around 20.

The reason for the high condition numbers is the correlation between the input factors for the compression ratio and the for the coatings. The pistons with the full coating have the lowest compression ratios, the pistons without TBC have the highest compression ratios, and the pistons with only top coatings have compression ratios in the middle of the range. This issue was investigated and taken into account during creation of the models by excluding factors and looking at the

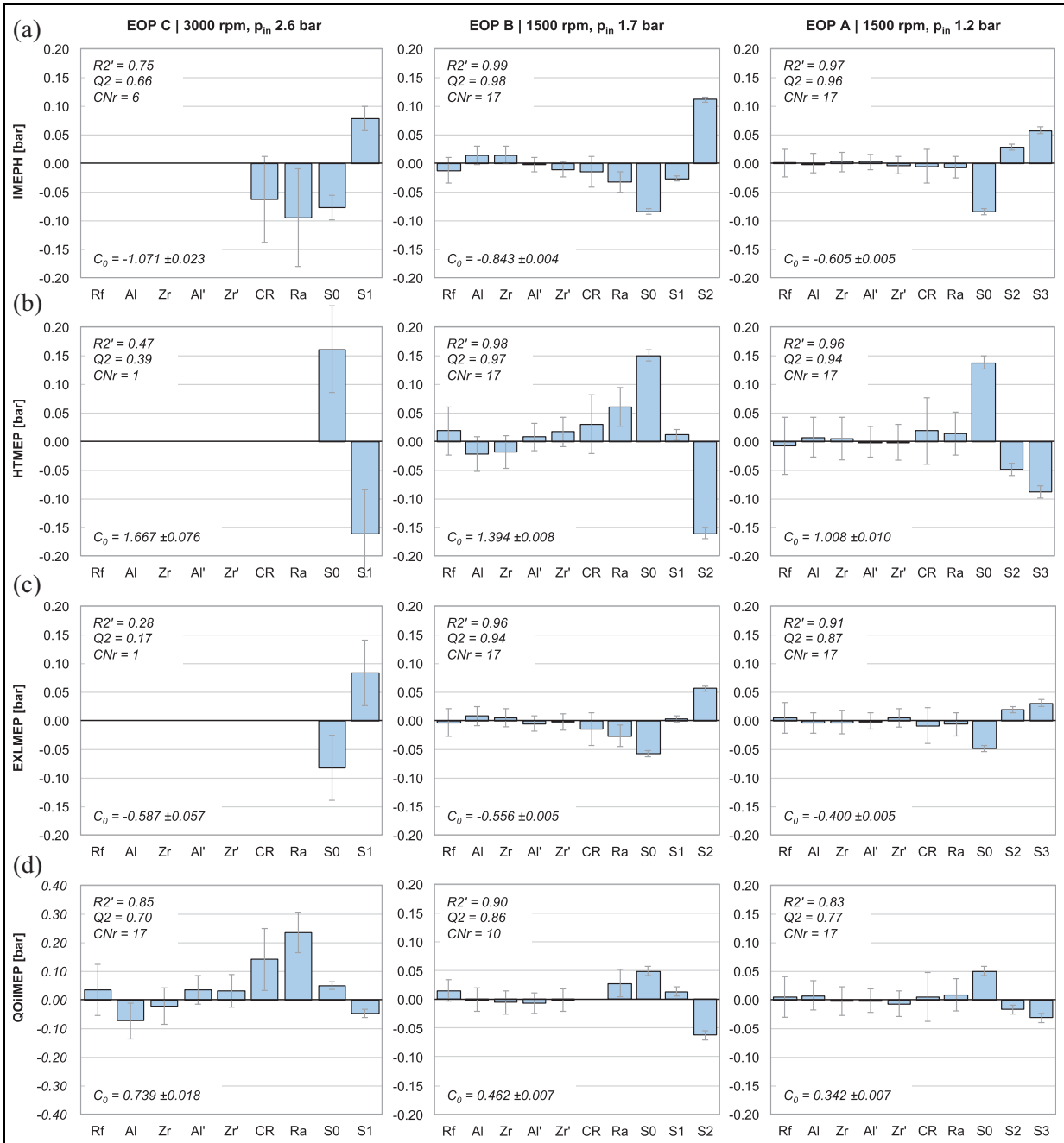


Figure 16. MLR model coefficients for the energy flow distribution under motored conditions. The error-bars represent the 95% CI. The legend for the coefficient labels can be found in Table 7.

consistency of the models for the different responses. The result of the collinearity is that the confidence levels for the factors are relatively low. However, the coefficients for compression ratio and coating show very similar trends when comparing the different models, for example, for ratHT and ratQOil, which puts some more confidence in the modeling results.

The equations below show how a response is calculated for the motored and fired engine results. In equation (14), the coefficient C_P is chosen for EOP C, coefficient C_E is used for EOP B and A

$$\begin{aligned} \text{response} = & C_0 + C_{TBC} + C_{CR} \times CR^* \\ & + C_{Ra} \times Ra^* + C_S \quad (\text{motored engine}) \end{aligned} \tag{13}$$

$$\begin{aligned} \text{response} = & C_0 + C_{TBC} + C_{AB} + C_{CR} \times CR^* \\ & + C_{Ra} \times Ra^* + C_P/C_E \quad (\text{fired engine}) \end{aligned} \tag{14}$$

$$\begin{aligned} \text{factor}^* = & 2 \times \frac{\text{factor} - \text{factor}_{\text{median}}}{\text{factor}_{\text{max}} - \text{factor}_{\text{min}}}; \\ & -1 \leq \text{factor}^* \leq 1 \end{aligned} \tag{15}$$

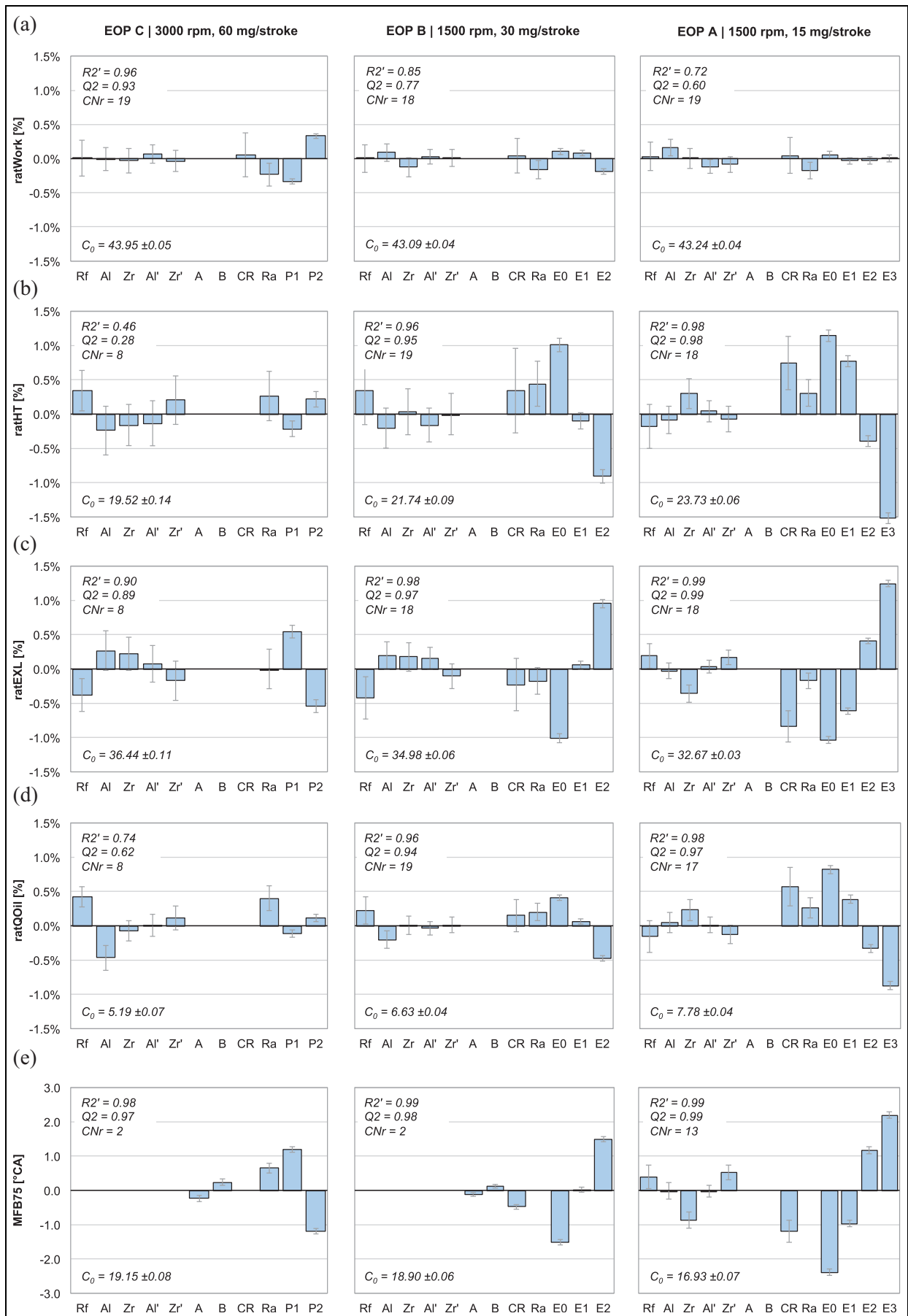


Figure 17. MLR model coefficients for the energy flow distribution and combustion phasing with fired engine operation. The errorbars represent 95% CI. The legend for the coefficient labels can be found in Table 7.

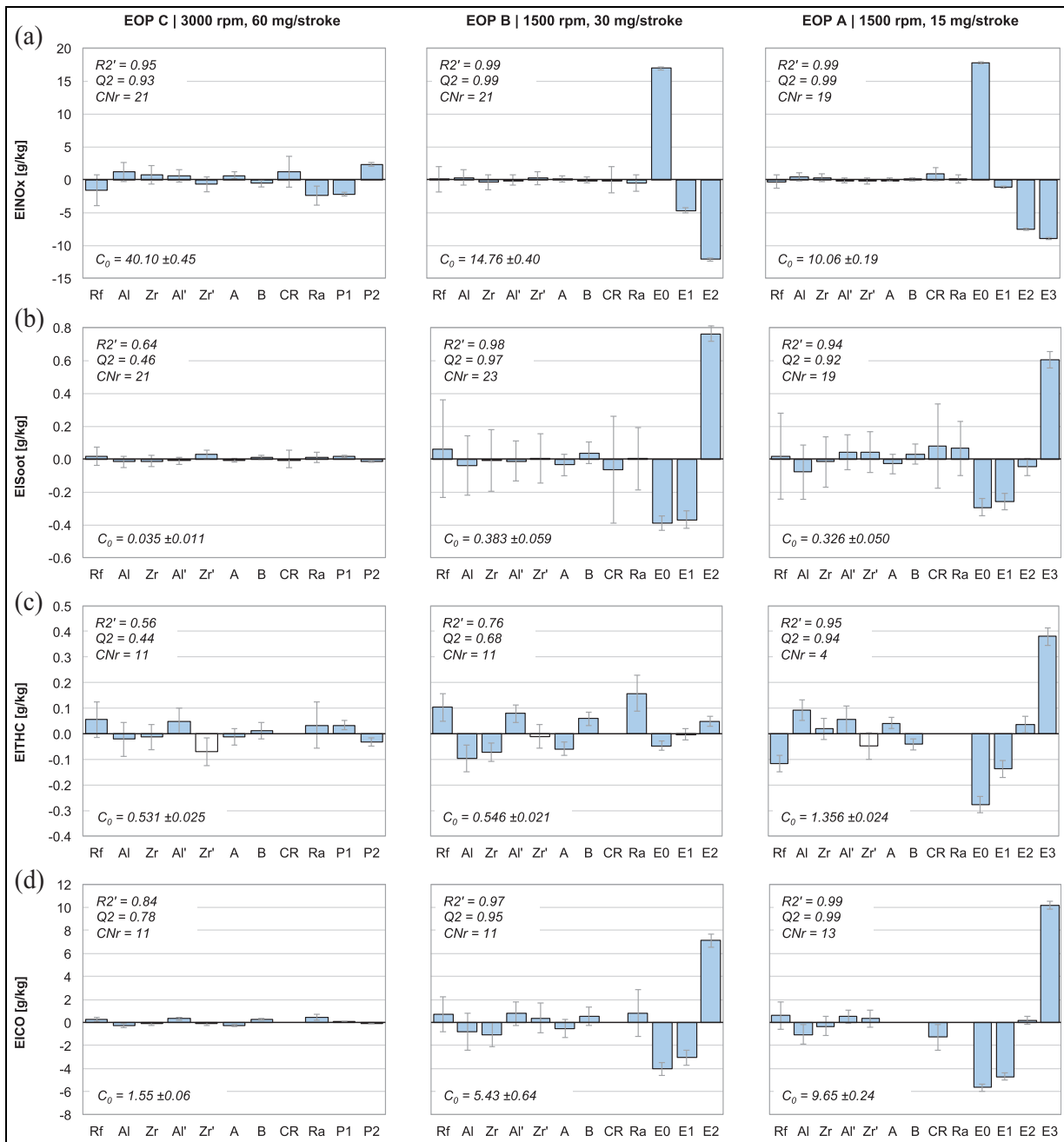


Figure 18. MLR model coefficients for the fuel-specific exhaust emissions of NO_x, Soot, THC, and CO. The error-bars represent the 95% CI. The legend for the coefficient labels can be found in Table 7.

Appendix 4

Results from the MLR model—fuel-specific energy distribution and position of MFB75

Results from the MLR model (fuel-specific energy distribution and position of MFB75) are shown in Figures 19 and 20.

Results from the MLR model—fuel-specific emissions

Results from the MLR model (fuel-specific emissions) are shown in Figures 21 and 22.

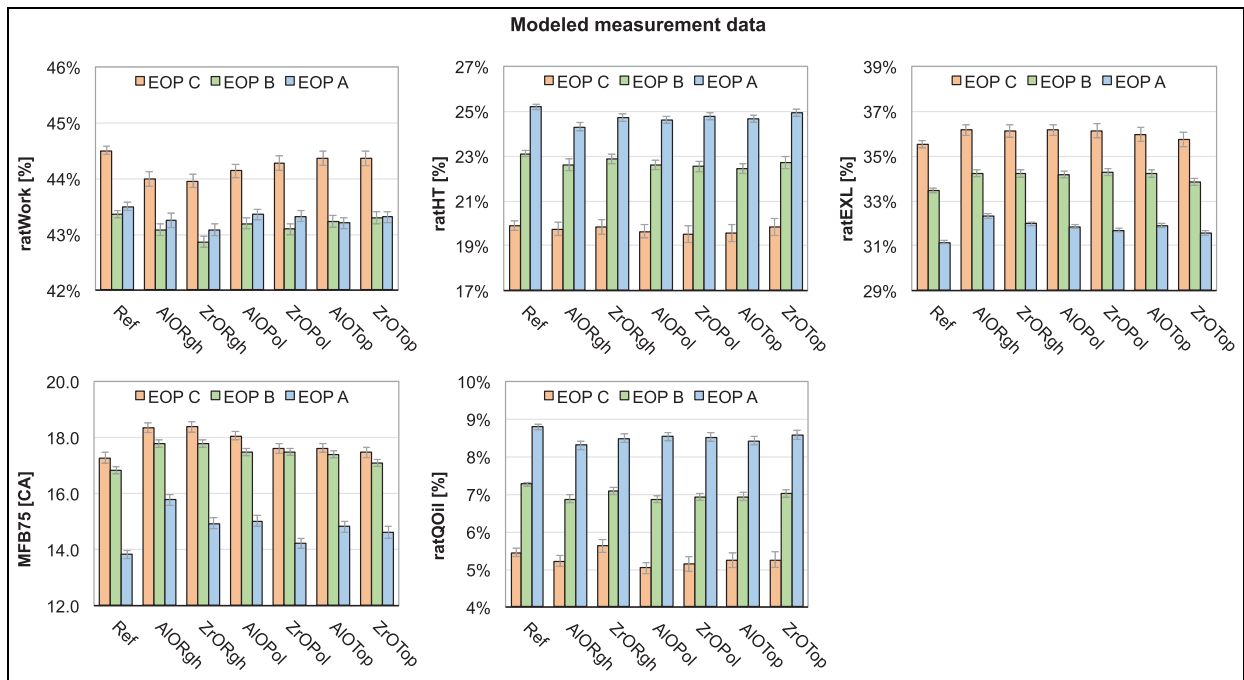


Figure 19. Modeled measurement data from the MLR model. Surface roughness R_a and CR for each piston variant as measured in Tables 3 and 4. Pfuel = P2, EGR = E0, representing run numbers 7, 12, and 20 in Table 9 in Appendix 2. The error-bars represent 95% CI.

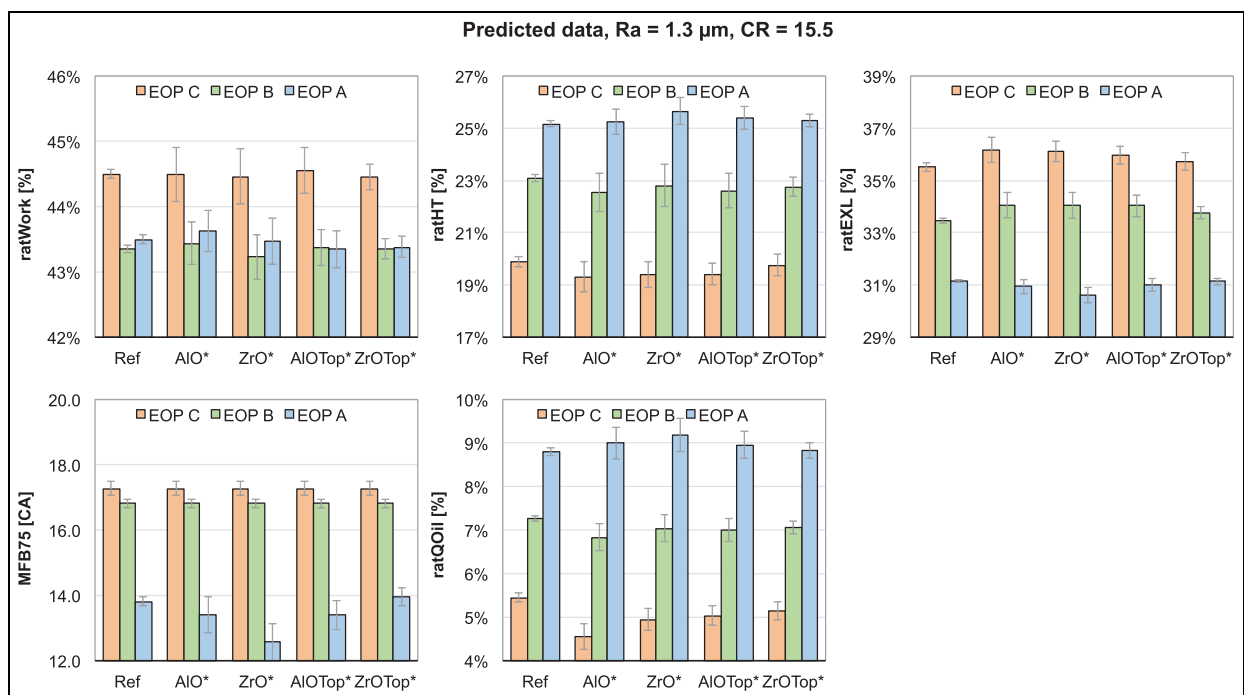


Figure 20. Predicted responses from the MLR model with $R_a = 1.3 \mu\text{m}$ and CR = 15.5 for all pistons. Pfuel = P2, EGR = E0, representing run numbers 7, 12, and 20 in Table 9 in Appendix 2. The (*) with the piston factor labels indicates predicted values. The error-bars represent 95% CI.

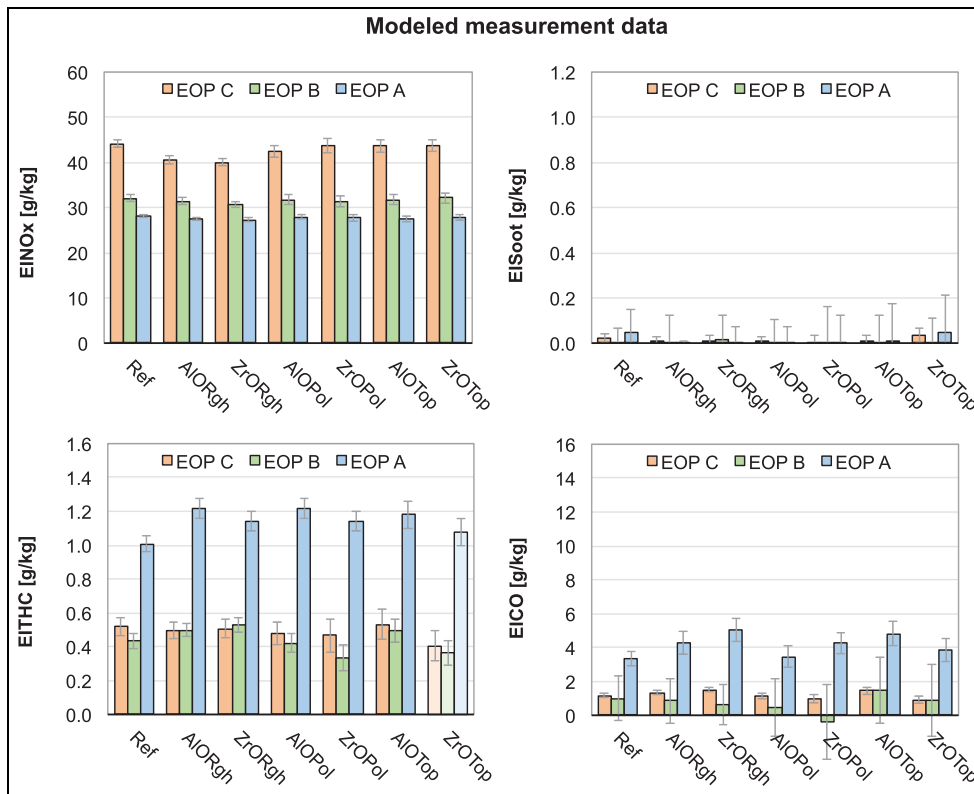


Figure 21. Modeled measurement data from the MLR model. Surface roughness R_a and CR for each piston variant as measured in Tables 3 and 4. Pfuel = P2, EGR = E0, representing run numbers 7, 12, and 20 in Table 9 in Appendix 2. The THC measurements for the ZrOTop piston had a small negative offset, indicated with the lighter shade color bars. The error-bars represent 95% CI.

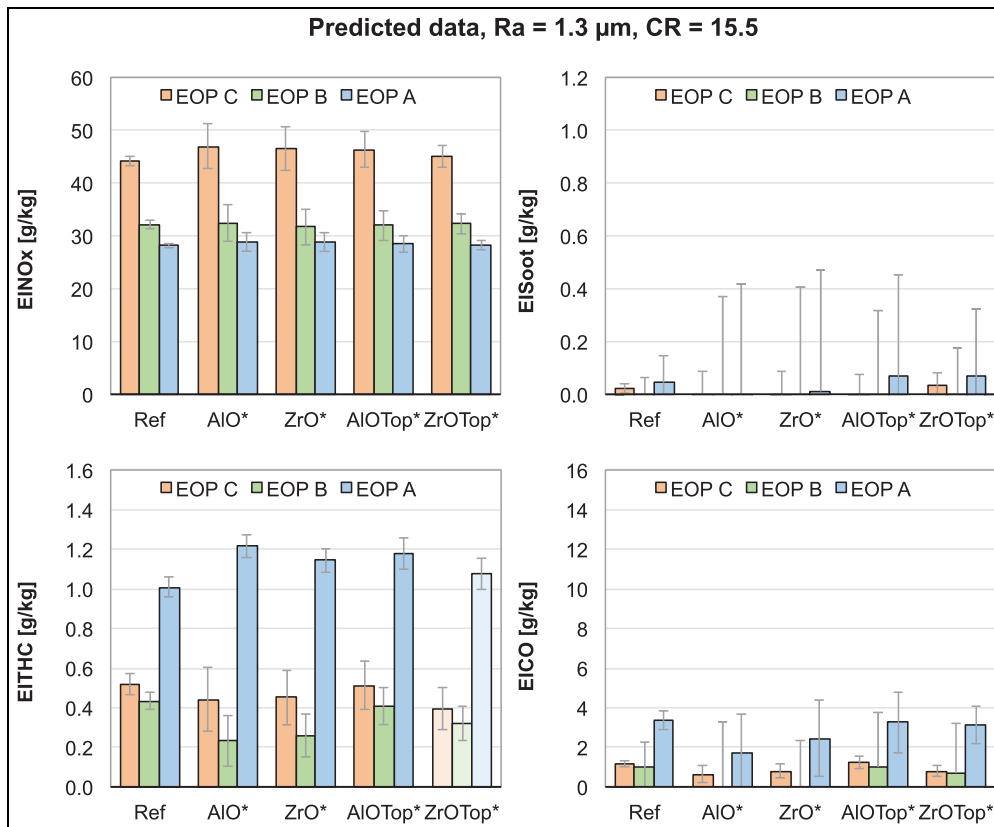


Figure 22. Predicted responses from the MLR model with $R_a = 1.3 \mu\text{m}$ and CR = 15.5 for all pistons. Pfuel = P2, EGR = E0, representing run numbers 7, 12, and 20 in Table 9 in Appendix 2. The THC measurements for the ZrOTop piston had a small negative offset, indicated with the lighter shade color bars. The (*) with the piston factor labels indicates predicted values. The error-bars represent the 95% CI.